

Assigned: Friday, November 30

Due: Friday, December 7

Note: Please hand in problems on separate sheets of paper

Note: The problems marked with “\*” are ones that will be useful to do as part of your Quiz preparation.

## Problem S8.1: Look-back to Lectures S10, S11, S12 (10 points)

The linearized longitudinal motion of a helicopter near hover, as shown in the figure below, can be modeled by the normalized third-order system

$$\underbrace{\begin{bmatrix} \dot{q} \\ \dot{\theta} \\ \dot{v} \end{bmatrix}}_{\dot{\vec{x}}} = \underbrace{\begin{bmatrix} -0.4 & 0 & -0.01 \\ 1 & 0 & 0 \\ -1.4 & 9.8 & -0.02 \end{bmatrix}}_A \underbrace{\begin{bmatrix} q \\ \theta \\ v \end{bmatrix}}_{\vec{x}} + \underbrace{\begin{bmatrix} 6.3 \\ 0 \\ 9.8 \end{bmatrix}}_B \gamma,$$

where the input is  $u(t) = \gamma(t)$ , the rotor tilt angle, and the other notation is defined in the figure.

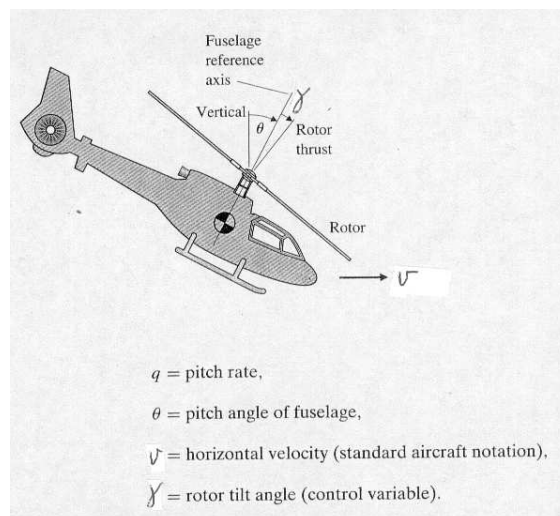


Figure 1: Sketch of helicopter hover dynamics system, Question S8.1. From Franklin, Powell and Emami-Naeini, *Feedback Control of Dynamics Systems*, Fifth edition, 2005.

- \*(a) Draw the block diagram for the system.
- \*(b) Determine, by hand, the characteristic equation of the system.
- (c) Compute the eigenvalues of the system (you can use Matlab to do this).

- (d) What do the eigenvalues tell you about the dynamics of the helicopter system?
- (e) We wish to change the dynamics of our helicopter, using a feedback control system. If our controlled system has eigenvalues  $\lambda_1 = -2$ ,  $\lambda_2 = -1 + j$ , and  $\lambda_3 = -1 - j$ , what should the characteristic equation of the controlled system be?

In Lectures S17 and S18, we will talk about ways to change the dynamics of a system using feedback control.

## Problem S8.2: Look-back to Lecture S13 (10 points)

\*(a) Prove the following properties of the state transition matrix  $\Phi(t)$ :

1.  $\Phi(0) = I$
2.  $\Phi^{-1}(t) = \Phi(-t)$
3.  $\Phi(t_2 + t_1) = \Phi(t_2)\Phi(t_1)$

(b) Verify Properties 1 and 2 for:

$$\dot{\vec{x}} = \begin{bmatrix} 0 & 1 \\ -5 & -2 \end{bmatrix} \vec{x}$$

## Problem S8.3: Look-back to Lectures S14, S15, S16 (10 points)

The state-space system

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -2 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$

$$\vec{y} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

has state transition matrix

$$\Phi(t) = \begin{bmatrix} 2e^{-t} - e^{-2t} & e^{-t} - e^{-2t} \\ -2e^{-t} + 2e^{-2t} & -e^{-t} + 2e^{-2t} \end{bmatrix}$$

- \*(a) Compute the output response of the system to a unit step input,  $u(t) = \sigma(t)$ , and initial conditions  $\vec{x}(0) = [1 \ 1]^T$ .
- \*(b) Is the system underdamped, overdamped or critically damped? Explain why.
- (c) Using Matlab, plot the output response computed in part (a).

Extra practice for the final: Derive the state transition matrix given above.